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Litipu Aihaiti, Shuxing Li, Dilare Halmurat, Jiajing Zhou, Siwei Li, Rong-Jun Xie

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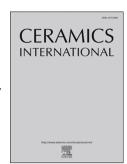
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# Significant enhancement on creep resistance of the near

# stoichiometric SiC fibers by ultrafast Joule heating

- 3 Litipu Aihaiti <sup>a</sup>, Shuxing Li <sup>a</sup>, Dilare Halmurat <sup>a</sup>, Jiajing Zhou <sup>a</sup>,
- 4 Siwei Li<sup>a, c\*</sup>, Rong-Jun Xie a, b\*\*
- 5 a College of Materials, Xiamen University, Xiamen, Fujian 361005, China
- 6 b State Key Laboratory of Physical Chemistry of Solid Surface, College of Materials, Xiamen
- 7 University, Xiamen 361005, China
- 9 Education, Xiamen 361005, China.
- \* Corresponding author at: Key Laboratory of High Performance Ceramic Fibers (Xiamen
- 11 University), Ministry of Education, Xiamen 361005, China. E-mail addresses:
- 12 swli@xmu.edu.cn (S. Li).

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- \*\* Corresponding author at: College of Materials, Xiamen University, Xiamen, Fujian 361005,
- 14 China. E-mail addresses: rjxie@xmu.edu.cn (R-J Xie).

### **ABSTRACT**

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- Heat treating is effective to enhance the thermal stability of the polymer derived
- 17 SiC fibers through increasing the grain size and purifying the grain boundary. However,
- a long-term heat treatment usually causes serious damages to the fibers and finally
- sacrifices the tensile strength. In this work, we apply an ultrafast (5-30 s) and high
- 20 temperature (1900-2100°C) heat treatment on the near stoichiometric SiC fibers
- 21 (namely C3) by Joule heating. As compared with the conventional heating techniques,
- 22 the ultrafast Joule heating yields evident grain growth but less fiber damage. At above
- 23 2000°C, an instantaneous decomposition of the surface SiC grains leading to grain size
- 24 reduction and a non-monotonically grain size distribution along the fiber diameter,
- 25 which also contributes to decrease the surface roughness. The C3 fibers heat treated at
- 26 2100°C for 15 s show excellent creep resistance when compared to some other
- commercial SiC fibers, and also keeps high tensile strength.
- 28 Keywords: Joule heating, SiC fiber, creep resistance, tensile strength, microstructure

### 29 1. Introduction

With the development of the aerospace industry, the hot section temperature of the

new generation aero engines is expected to reach 1400-1700°C, surpassing the 1 survivability limit of nickel-based high temperature alloys (which can withstand up to 2 3 1100°C) [1, 2]. SiC fiber-reinforced SiC ceramic matrix composites (SiC<sub>f</sub>/SiC CMCs), however, have advantages that satisfy the dual requirements of high strength and high-4 temperature resistance, which have the working temperature more than 200°C higher 5 6 than the nickel-based alloys [3]. At the same time, the SiC<sub>f</sub>/SiC CMCs are much lighter than metal alloys, higher oxidation resistance than carbon materials, and higher 7 8 toughness, and reliability than ceramic materials. Furthermore, they have good 9 corrosion resistance, wear resistance and microwave-absorbing abilities. Therefore, they have a variety of applications in fields of nuclear fusion, aerospace and high-10 11 temperature radar stealth technologies [4-6]. The general processing route from fibers to composites consists of several steps, 12 such as the preparation of fiber preforms, introduction of interphases around fibers, and 13 the matrix formation (densification) [7]. The thermo-mechanical properties of 14 composites are greatly determined by the fibers, it is thus of great significance to 15 16 prepare SiC fibers with better mechanical and thermal properties. SiC fibers were first synthesized by Yijima at the laboratory in 1974, and industrialized by Nippon Carbon 17 in Japan in 1980 [6, 8]. Since then, some commercial SiC fibers have been developed, 18 with the trademarks of Nippon, Tyranno, Sylramic, and etc. These products can be 19 divided into three generations based on the composition and microstructure of fibers. 20 Among them, the third generation SiC fibers are near chemistry stoichiometric, which 21 22 therefore have better mechanical properties and higher thermal stability [9, 10]. 23 Although the SiC fibers can retain good strength at high temperatures, they 24 experience severe creep that will lead to the failure of fibers and even composites. The 25 creep becomes more noticeable at temperatures higher than about 45% of the absolute melting point of ceramics [11]. Creep under high temperature and low load is primarily 26 27 induced by the diffusion of atoms through grain boundaries and within the crystal lattice. 28 This type of creep exhibits a high sensitivity to the grain size and the purity of grain

boundaries. [12]. Among the three generations of SiC fibers, third generation ones

possess the best creep resistance at high temperatures due to their near chemistry

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stoichiometric nature, larger grain size, and higher grain boundary purity [8, 11, 13]. 1 For example, the grain size obtained by TEM analyses is the sequence of Tyranno SA 2 3 (~200 nm) > Sylramic-iBN (>100 nm) > Sylramic (~100 nm) > Hi-Nicalon Type SA (HNLS) (~20 nm) [14, 15], which largely depends on the sintering temperature. 4 However, the creep resistance behaviors is the sequence of Sylramic-iBN > HNLS > 5 6 Tyranno SA > Sylramic [16]. Although the Tyranno SA and Sylramic fibers have a bigger grain size than HNLS, their creep resistance is lower due to the low purity of 7 8 their grain boundaries [8]. In short, the larger grain size and purer grain boundaries 9 endow SiC fibers with higher creep resistance. 10 High-temperature heat treatment is an effective way to improve the creep resistance of SiC fibers, which can increase the grain size and purify the grain boundary. 11 Sha et al. [17] investigated the influence of heat treatment temperature on the creep 12 behavior of three kinds of commercialized SiC fibers (i.e., Hi-Nicalon, HNLS, and 13 Tyranno-SA) under an Ar atmosphere, and found that the creep resistance of all SiC 14 fibers was improved by heat treatment, which is mainly related to the grain size of β-15 16 SiC and the composition at or adjacent to the grain boundary. However, the heat treatment also reduces the strength of SiC fibers. The strength reduction is ascribed to 17 the grain growth, defect formation and residual thermal stress [18]. Dicarlo et al. [19] 18 19 treated the Sylramic SiC fibers at 1800°C in N<sub>2</sub> to obtain the Sylramic-iBN fibers. The Sylramic-iBN fibers show better creep resistance and thermal stability than the original 20 ones due to the grain coarsening and grain boundary purification. Moreover, the heat-21 22 treated fibers retain a relatively high strength on account of the surface protection by 23 boron nitride (BN). The formation of BN is attributed to the migration of excess boron 24 to the surface followed by its reaction with N<sub>2</sub>. Gao et al. [20] studied the thermal 25 behavior of second-generation KD SiC fibers in N2 and Ar, and found that the heat treatment caused much larger strength degradation in Ar. The strength degradation is 26 attributed to the grain coarsening and porous surface of the fibers, and the 27 28 decomposition of SiC<sub>x</sub>O<sub>y</sub> and SiC<sub>x</sub>N<sub>y</sub>O<sub>z</sub> phases occurs more easily in Ar. 29 The heating rate also plays an important role in affecting the thermal behavior of

SiC fibers. Shimoo et al. reported that the pyrolysis of SiC fibers at a faster heating rate

(e.g., 600°C/h) is beneficial to improving their strength [21]. Kim et al. [22] investigated 1 2 the effects of the pyrolysis temperature and heating rate on the mechanical properties 3 of SiC fibers, and addressed that a faster heating rate (e.g., 40°C/min) resulted in higher tensile strength and greater elasticity at room temperature. In our previous work, we 4 performed rapid heat treatment on the BN-coated near stoichiometric SiC fibers at 1800°C 5 6 for 60 seconds in Ar, and found that the creep resistance of the treated SiC fibers was better than that of the untreated fibers and even higher than that of the commercialized 7 HNLS SiC fibers. Furthermore, it keeps excellent strength compared to the as-prepared 8 9 one after being treated at 1500°C for 1 hour. These good performances of the treated 10 SiC fibers are ascribed to the moderate grain growth and smooth surface after fast heat treatments. However, the influence of even higher temperature fast heat treatment and 11 different holding times on the microstructure, morphology and mechanical properties 12 13 of near stoichiometric SiC fibers has not been discussed systematically .[23].

In this work, we apply an ultrafast high temperature heat treatment on SiC fibers

and investigate their structural evolution and mechanical properties.

### 2. Experimental section

#### 2.1. Materials

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The near chemistry stoichiometric SiC fibers used in this study were the same as those used in our previous work, which were commercially available from LEADASIA Co. Ltd. (Quanzhou, Fujian, China) and abbreviately named as C3 [23]. Some basic properties of the C3 fibers can be found in our previous work. The fiber is supposed to possess an oxygen content of 0.8% and a C/Si ratio of 1.07. Its density is measured as 3 g/cm<sup>3</sup>. A carbon-rich layer with a thickness of 40 nanometers is formed on the fiber surface. The fiber exhibits a strength of 4.1 GPa and a Young's modulus of 336 GPa. The grain size of the C3 fiber by TEM is 14.5 nm. They were similar to those of commercialized Hi-Nicalon Type S (HNLS) produced by Nippon Carbon Inc. [24].

### 2.2. Heat treatment

A single C3 fiber was placed between two flexible sheets of carbon paper, and the end of the carbon paper was connected to two electrodes of the Joule heating equipment

- 1 (JSJ150-II, Hefei, China). The C3 fibers were heat treated at varying temperatures (i.e.,
- 2 1900, 2000 and 2100°C) for different holding times (i.e., 5, 15 and 30 seconds) under
- 3 the nitrogen (99.999% in purity) atmosphere. The temperature was measured by an
- 4 infrared high-temperature thermometer (CK-7030A-MJ, Xi'an, China).

#### 2.3. Characterizations

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- The field emission scanning electron microscopy (SEM, SU-70, Hitachi, Japan)
- and transmission electron microscopy (TEM, JEM-2100F, JEOL, Japan) were used in
- 8 combination with energy dispersive spectroscopy (EDS) for microstructural
- 9 observations and compositional determination of SiC fibers. Samples for TEM
- observations were prepared by a focusing ion beam system (FIB, NB5000, Hitachi,
- Japan). The phases of fibers were identified by an X-ray diffractometer (XRD, D8-A25,
- 12 Bruker-Axs, USA).
- In addition, the grain size of SiC in the fibers was observed and then statistically
- measured based on the high-resolution transmission electron microscopy (HRTM)
- images. Grain size measurements were performed manually using the ImageJ software
- 16 (more than 40 particles were counted and then averaged).

### 2.4. Mechanical properties test

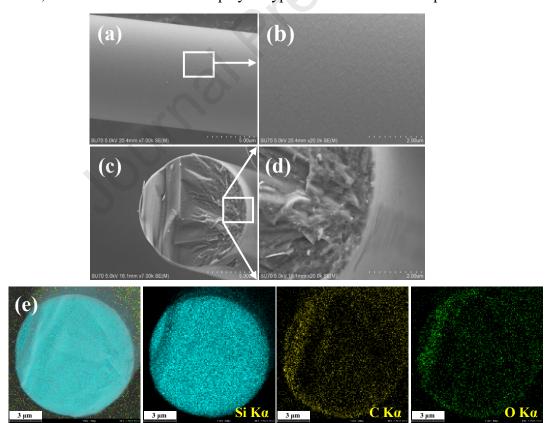
- The high-temperature creep resistance of the fiber was determined by a modified
- bending stress relaxation (BSR) method that was used in our previous work [23, 25].
- 20 The modification was to fill the slit formed between the graphite clamp and graphite
- 21 mandrel with carbon powders before placing the mandrel into the tube furnace to further
- 22 minimize the oxidation degree of fibers under high temperatures. The oxidation may
- have a little impact on the measurement accuracy of creep resistance of the fibers. In
- 24 this method, the creep resistance was evaluated by the stress relaxation ratio (m), which
- is defined as  $m = 1 R_0/R_a$ , where  $R_0$  is the radius of the graphite mandrel and  $R_a$  is the
- residual radius of the fiber after creep. The SiC fiber tow containing 10 single fibers
- 27 were bent around a graphite mandrel with a fixed radius ( $R_0 = 2$  mm) and held at
- constant strain. The mandrel was placed into a graphite clamp, and sprinkled very fine
- 29 carbon dust between the mandrel and creep until the slit between them was filled, and
- 30 then the graphite clamp was put into a tube furnace. Finally, they were heated at 1200,

- 1 1300, 1400 and 1500°C for 1 hour under 99.999% high purity argon gas, respectively.
- 2 After cooling to the ambient temperature, the fibers were removed from the mandrel,
- and the residual radius  $R_a$  of the fiber loop was measured. The value of m was averaged
- 4 over ten specimens in each test.
- 5 The tensile strength and elastic modulus of single fibers with a gauge length of 25
- 6 mm are examined by using an Instron-type test machine (Microtester-5948, England).
- 7 The values of tensile strength and elastic modulus of each sample were averaged over
- 8 at least 25 filaments.

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#### 3. Results and discussions

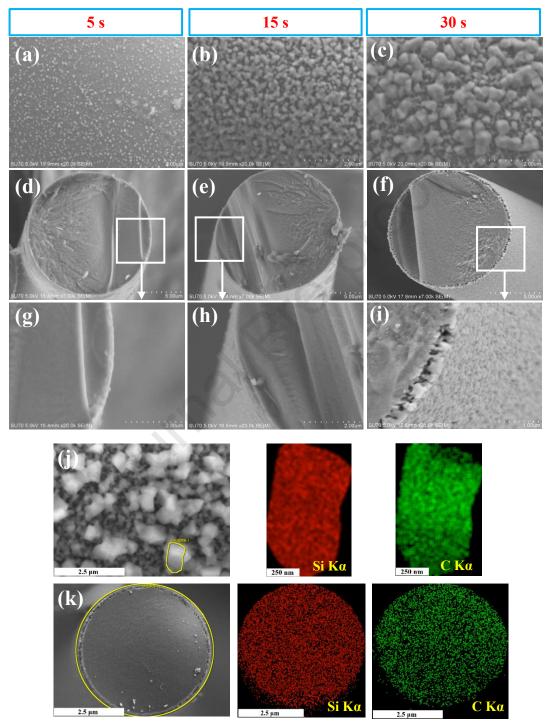
- Fig. 1 shows the morphology and Energy dispersive spectroscopy (EDS) of the asprepared SiC fiber (C3). C3 shows a smooth and dense surface containing small
- particles, and its fracture surface displays a typical brittle nature. The presence of



13 Fig. 1. Scanning electron microscopy (SEM) micrographs of the surface (a, b) and

- fracture morphology (c-d) of as-prepared C3 fiber. Energy dispersive spectroscopy
- 15 (EDS) of C3 fiber (e).

- 1 oxygen can be detected from the EDS in Fig. 1e.
- After heat treatment at 1900°C, the island-like particles are produced on the
- 3 surface. These particles come together over time to form a discontinuous layer that



4 Fig. 2. Scanning electron microscopy (SEM) micrographs of the surface (a-c) and

- 5 fracture morphology (d-i) of C3 fiber treated at 1900 °C for different times. (a, d, g) 5s;
- 6 (b, e, h) 15s; (c, f, i) 30s; Energy dispersive spectroscopy (EDS) of C3 fiber treated at
- 7 1900°C for 30 s. (j) Surface scan; (k) Cross section scan.

- looks like a fiber skin. Composition of these particles are mainly SiC, which can be 1
- verified by the EDS surface scanning images (Fig. 2j and 2k). These SiC particles grow 2
- 3 more likely through the gas-phase reaction mechanism with the following equations
- [26, 27]: 4

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$$SiO(g) + 3CO(g) \rightarrow SiC(s) + 2CO_2(g)$$
 (1)

At higher temperatures (i.e., 2000 and 2100°C), pores instead of small particles 7 are formed on the surface of the fiber. At the same time, from the cross-sectional

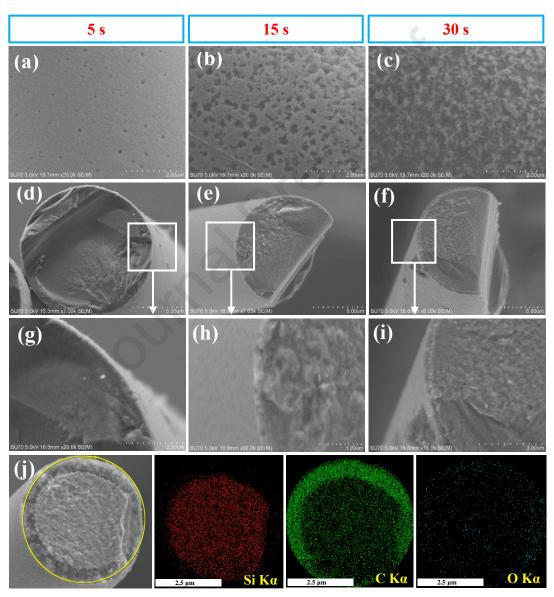
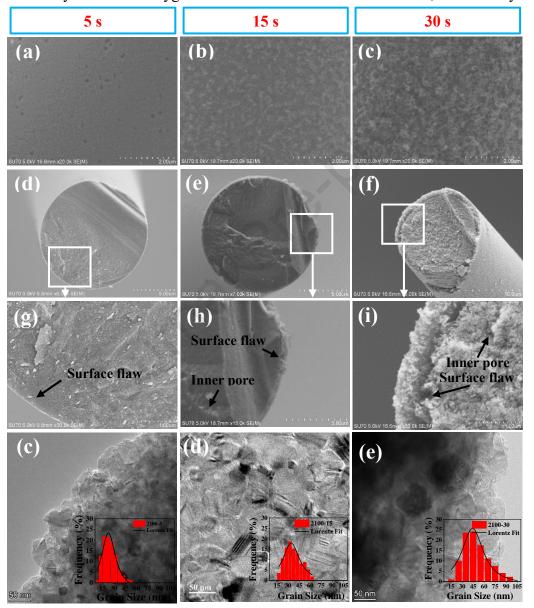


Fig. 3. Scanning electron microscopy (SEM) micrographs of the surface (a-c) and fracture morphologies (d-i) of the C3 fiber treated at 2000°C for different times: (a, d, g) 5 s; (b, e, h) 15 s; (c, f, i) 30 s; Energy dispersive spectroscopy (EDS) of the C3 fiber treated at 2000°C for 30 s (j).

- image of the fiber, it can be clearly observed that the fiber surface is covered by a cortex,
- 2 which we called the core/shell structure. The size and number of pores as well as the
- 3 thickness of the shell on the fiber increase with time, as shown in Figs. 3 and 4. The
- 4 composition of shell is mainly carbon and the core is mainly SiC, as given by the energy
- 5 dispersive spectroscopy (EDS) in Fig. 3j. Meanwhile, the existence of a certain amount
- of uniformly distributed oxygen is observed from the EDS. However, the intensity of



7 Fig. 4. Scanning electron microscopy (SEM) micrographs of the surface (a-c) and

- 8 fracture morphologies (d-i) of the C3 fibers treated at 2100°C for different times: (a, d,
- 9 g) 5 s; (b, e, h) 15 s; (c, f, i) 30 s. (c, d, and e) Grain size distribution of the SiC fibers
- heat treated at 2100°C for different durations: (c) 5s; (d) 15s; (e) 30s.

- the oxygen signal is relatively lower than that of the as-prepared sample (as shown in
- 2 Fig. 1e), which indicates a decrease in oxygen content after high-temperature heat
- 3 treatment. Carbon layer exhibits a porous structure, while the interior of SiC fiber is a
- 4 relatively dense structure. Surface carbon layer is formed by the decomposition of
- 5 surface SiC grains at ultra-high temperatures, as given by the following reaction [30,
- 6 31]:

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$$\operatorname{SiC} \rightarrow \operatorname{Si}(g) + \operatorname{C}(s)$$
 (2)

- Although, the  $\Delta G$  of this reaction is >0 at those temperature ranges, however, this
- 9 reaction is likely to proceed due to the effect of impurities (i.e., small amounts of
- oxygen, free carbon, and amorphous phase) in the SiC fibers, and more importantly
- the presence of high surface-active SiC nano-grains, which weaken the Si-C bond,
- causing SiC to decompose at low temperatures [30-32].
- However, there are no significant changes in the fracture morphology of the fiber,
- except for the grain coarsening and the pore formation in the cross section when the
- 15 fiber is treated at 2100°C for 30 s (Fig. 4f and Fig. 4i). The defects are produced between
- the SiC fiber and carbon skin are probably due to the difference in the thermal
- expansion coefficient of the SiC fiber and carbon (4.  $5\times10^{-6}$ /°C vs 2.0-3.0×10<sup>-6</sup>/°C) [26].
- In addition, as can be seen from Fig. 4c-e, the grain size of the SiC fibers heat-treated
- at 2100°C increased monotonically with the treatment durations. Specifically, the grain
- sizes are 24.26, 36.83, and 49.11 nm for the duration of 5s, 15s, and 30s, respectively.
- 21 The high-temperature creep resistance of the C3 fibers, conducted at 1300°C for
- 22 1h, was verified by using the bend stress relaxation (BSR) method and evaluated by the
- 23 stress relaxation parameter (m), as shown in Fig. 5a and Table 1. In this method, the
- creep resistance of the fibers increases with the m value increasing from 0 to 1. The m
- value of  $\sim 1$  indicates the best creep resistance performance, whereas that of  $\sim 0$  means
- 26 the worse creep resistance of the fiber.
- 27 After heat treatment at high temperatures, all annealed samples show higher m
- values. In addition, the m value increases with the annealing temperature. On the other
- 29 hand, with increasing the heat treatment time, the m value monotonically increases for
- 30 the sample annealed at 1900°C, while it firstly increases and then remains almost

unchanged for the samples treated at 2000°C and 2100°C.

Specially for the SiC fiber heat-treated at 2100°C, the grain size increases with time, as depicted in Fig. 4.c-e, while the value of m remains essentially unchanged. For the SiC fiber heat-treated at 2100°C for 15 seconds, although its grain size is larger than that of the fiber heat-treated for 5 seconds, it maintains the same m value as the latter. Table 1 Bend stress relaxation parameters (m) of the treated C3 fibers crept at 1300°C for 1h

Heat trea temperatur		Bend stress relaxation ratio n (Heat treatment period)		
(°C)	0s	5s	15s	30s
1900	0.6518	0.7196	0.7295	0.7940
2000	0.6518	0.8411	0.8385	0.8377
2100	0.6518	0.8635	0.8555	0.8655

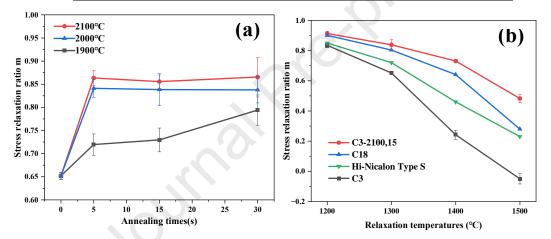


Fig. 5 (a) Bend stress relaxation parameters (m) of the heat treated C3 fibers crept at 1300°C for 1h. (b) Bend stress relaxation parameters (m) of four type fibers crept at different temperatures for 1 h.

This is most likely related to the formation of a carbon cortex on the fiber surface as shown in Figure 4h. The carbon cortex reduces the cross-section of the SiC fiber. Consequently, the SiC fiber bears a greater load per unit cross-sectional area during creep strain than the fiber with a thermal duration of 5 seconds. Moreover, the fiber with a duration of 30 seconds has an even larger grain size than those with durations of 5 and 15 seconds. However, it maintains the same m values as them. This is most probably due to its cavity structure as shown in Figs. 4f and 4I, which is favorable for grain boundary sliding. In short, it seems that the temperature of 2000°C is a threshold

one for activating a fast grain growth of the SiC fibers.

Figure 4(b) shows the bend stress relaxation parameters of four SiC fibers crept for 1h at different temperatures. The data of Hi-Nicalon Type S (HNLS) and C18 (C3 heat treated at 1800°C for 60 s) are derived from the literature [12]. The C3-2100,15 fiber, heat treated C3 at 2100°C for 15 s, has the maximal value among the tested fibers at all temperatures. High temperature creeping is mainly caused by the grain boundary migration that is sensitive to the grain size and grain boundary purification. The C3-2100,15 fiber exhibits larger grains and purer grain boundaries than the C18 and HNLS ones, showing the largest creep resistance among the four type fibers.

To investigate the influence of heat treatment time on the strength of SiC fibers, we measured the tensile strength of the fibers heat treated at 2100°C for varying times (Fig. 6a). One can see that the tensile strength of the fibers reduces monotonically with prolonging heat treatment time, and the strength retention of the fibers is 42.41, 37.73 and 15.47% for the heat treatment time of 5, 15 and 30 s, respectively. Increase the number and sizes of surface and inner defects with prolong the heat-treated times are responsible for the reduction in the tensile strength of the fibers. Although the strength of the C3 fiber decreases after the ultra-fast heat treatment, this result is still better than those reported in the literature [18], where the creep resistance of the HNLS SiC fiber is improved a little bit, but its strength is completely lost.

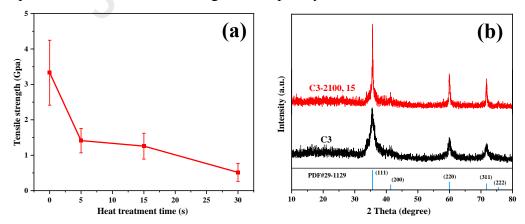
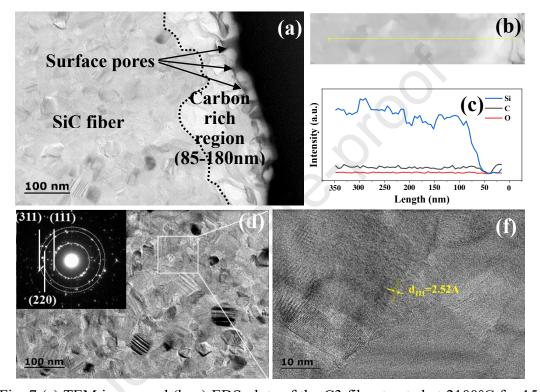


Fig. 6 (a)Tensile strength of the C3 fiber treated at 2100°C for different times. (b) XRD patterns of the C3 fiber and C3 fiber heat treated at 2100°C for 15 s (*i.e.*, C3-2100,15).

To gain a deep understanding why the rapid heat treatment at high temperature can improve the creep resistance of the fiber while keep relatively high tensile strength, the

- 1 microstructure and composition of the C3 fiber and the fibers treated at 2100°C for 15
- s were investigated. As shown in Fig. 6b, the diffraction peaks at 35.6°, 41.2°, 60.2°,
- $72.1^{\circ}$  and  $75.5^{\circ}$  assigned to the (111), (200), (220), (311) and (222) crystal planes of β-
- 4 SiC can be observed for both of the C3 and the heat-treated fibers, respectively.
- 5 Compared with the C3 fiber, the diffraction peak strength of the heat-treated fiber is
- 6 stronger, and the peak width is narrower, which implies an enhancement in crystallinity.



- 7 Fig. 7 (a) TEM image and (b, c) EDS plots of the C3 fiber treated at 2100°C for 15 s.
- 8 Selected area electron diffraction (SEAD) (d) and high-resolution transmission electron
- 9 microscope (HRTM) (f).

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As given in Fig. 7a, some pores are formed on the surface of the fiber after heat treatment at  $2100^{\circ}$ C for 15s. These pores are caused by the decomposition and oxidations of  $SiC_xO_y$  and SiC phases on the surface of SiC fibers. The presence of these pores leads to the strength degradation. In addition, the carbon-rich region is formed by the decomposition of surface SiC grains and the evaporation of Si after high-temperature treatment, as discussed above. The diffraction rings are assigned to the (111), (220) and (311) crystal planes of  $\beta$ -SiC (Fig. 7c), which is consistent with XRD

results. The spacing between the two planes of (111) is 2.52 Å (Fig. 7d).

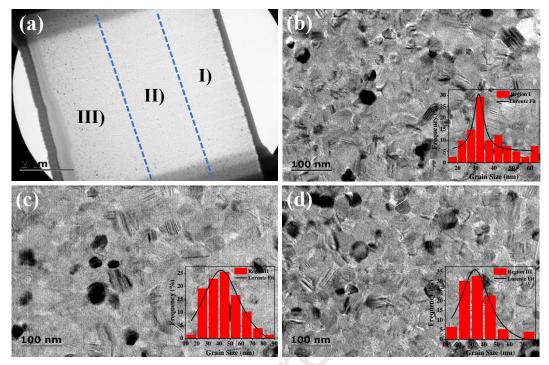


Fig. 8. TEM cross section images of the C3 fiber heat treated at 2100°C for 15 s (a) and the grain size distribution in three regions: (b) region I, (c) region II and (d) region III.

The size of SiC grains was measured from the TEM micrographs (Fig. 8). A thin sheet with a thickness of 20 nm and depth of 6.62 µm was sliced from the C3 fiber heat treated at 2100°C for 15 s using FIB, and was then divided into three regions from the surface to the core (*i.e.*, I, II and III), respectively. It is seen that the average grain size of SiC in the fiber from surface to core is 37.08 nm (region I), 40.70 nm (region II) and 35.68 nm (region III), respectively. The SiC grains in the middle region have the largest size, which is different from that reported in the literature [23, 33], where the grain size of SiC near the fiber surface is the biggest after heat treatment. Theoretically, the grain size near the surface should be larger than those far away from the surface due to the temperature gradient. The unusual grain growth observed in this study is most probably due to the decomposition of SiC grains near the surface is much faster than the grain growth under the ultrafast heat treatment, leading to smaller grains near the surface. Instead, the decomposition of SiC grains in the middle region (region II) is less than that on the surface, therefor this region exhibits larger grain size than the other two regions. The average grain size (37.82nm)

in the entire area of the fiber heat treated at 2100°C for 15s is approximately twice of the average grain size (*i.e.*,14.5 nm) of the as-prepared C3 fiber [20].

Based on the above discussion, it can be concluded that the SiC fiber with nearchemistry stoichiometry after ultrafast heat treatment exhibits higher creep resistance, larger grain size, and greater crystallinity (which can be presumed to have higher grain boundary purity) compared to the as-prepared one. Additionally, it still maintains a relatively high tensile strength. The enhanced creep resistance of the heattreated fiber can be ascribed to the enlarged grain size of SiC and improved purity at grain boundary. The relatively high strength of the SiC fiber after ultrahigh temperature heat treatment probably related to these factors: On the one hand, the shortening heat treatment period reduces the decomposition degree of the SiC<sub>x</sub>O<sub>y</sub> phases, which in turn reduces the surface and inner defects of the SiC fibers. In addition, the restriction of the SiC grain growth and formation of the carbon on the surface of SiC fiber helps to minimize and even to heal the surface defects. All of these results contribute to maintain the fiber strength. It can be predicted that the ultrafast heat treatment process introduced in this study is applicable to other types of SiC fibers, such as Tyranno and Sylramic fibers to improve the creep resistance while maintaining a high strength.

### 4. Conclusions

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- The ultrafast high-temperature heat treatment was conducted on the near-stoichiometric SiC fibers (*i.e.*, C3), and their microstructure and mechanical properties were investigated. The main results are summarized as follows:
  - (1) The size of SiC grains and voids in the fiber increases with elevated temperatures and prolonged treating time, resulting in a decrease in tensile strength. The ultrafast heat treatment at 2100°C for 15s activates an instantaneous decomposition of SiC grains in the near surface, leading to a non-monotonically grain size distribution along the fiber diameter.
  - (2) According to the BSR test results, all annealed fibers show higher creep resistance than the as-received one due to the grain coarsening and grain boundary

- purification. With increasing the treating time from 5 s to 30 s, the creep resistance
- 2 monotonically enhanced for the sample annealed at 1900°C, while it firstly
- enhances and then remains almost unchanged for the samples treated at 2000°C
- 4 and 2100°C.
- 5 (3) The C3 fiber heat treated at 2100°C for 15 s have a better creep resistance than
- 6 HNLS SiC fiber. At the same time, it maintains a relatively high strength (37.73%
- of the initial strength of 3.33GPa) due to the suppression on formation of surface
- 8 defects.

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### **Declaration of competing interest**

- The authors declare that they have no known competing financial interests or
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- 17 paper.

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Table 1 Bend stress relaxation parameters (m) of the treated C3 fibers crept at 1300°C for 1h

Heat treatment		Bend stress relaxation ratio m			
temperature		(Heat treatment period)			
(°C)	0s	5s	15s	30s	
1900	0.6518	0.7196	0.7295	0.7940	
2000	0.6518	0.8411	0.8385	0.8377	
2100	0.6518	0.8635	0.8555	0.8655	

## **Declaration of Interest Statement:**

The authors declare that they have no known competing financial interests or personal relationships that could have appeared to influence the work reported in this paper.